

Ground and Flight Testing of Non-Chrome Paint Systems; Part 1– Acceleration Factors

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Battelle-Columbus

Air Force Corrosion Managers Conference

Robins AFB, GA.

16-18 August 2011

including suggestions for reducing	ompleting and reviewing the collect this burden, to Washington Headqu uld be aware that notwithstanding ar DMB control number.	arters Services, Directorate for Infor	mation Operations and Reports	, 1215 Jefferson Davis	Highway, Suite 1204, Arlington	
1. REPORT DATE AUG 2011	2. REPORT TYPE			3. DATES COVERED 00-00-2011 to 00-00-2011		
4. TITLE AND SUBTITLE			5a. CONTRACT NUMBER			
Ground and Flight Acceleration Facto	; Part 1-	5b. GRANT NUMBER				
Acceleration Facto	5c. PROGRAM ELEMENT NUMBER					
6. AUTHOR(S)		5d. PROJECT NUMBER				
		5e. TASK NUMBER				
			5f. WORK UNIT NUMBER			
	ZATION NAME(S) AND AE ,505 King Avenue,C	8. PERFORMING ORGANIZATION REPORT NUMBER				
9. SPONSORING/MONITO	RING AGENCY NAME(S) A		10. SPONSOR/MONITOR'S ACRONYM(S)			
				11. SPONSOR/M NUMBER(S)	ONITOR'S REPORT	
12. DISTRIBUTION/AVAII Approved for publ	LABILITY STATEMENT ic release; distributi	on unlimited				
	otes 111 Air Force Corro deral Rights Licenso		d 16-18 Aug 2011	at Robins A	FB, GA. U.S.	
14. ABSTRACT						
15. SUBJECT TERMS						
16. SECURITY CLASSIFIC	17. LIMITATION OF ABSTRACT	18. NUMBER OF PAGES	19a. NAME OF RESPONSIBLE PERSON			
a. REPORT unclassified	b. ABSTRACT unclassified	c. THIS PAGE unclassified	Same as Report (SAR)	16	RESI UNSIBLE FERSUN	

Public reporting burden for the collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and

Report Documentation Page

Form Approved OMB No. 0704-0188



Basis of Study

- Provide Supporting Data on Performance of Wide Range of Non-Chrome or Reduced Chrome Paint Stackups for Decisions Regarding Chrome Elimination
- Evaluate Wide Range of Chemical Stresses/Threats To Coatings
- Emphasis on Corrosion In Severe Ground and Flight Environments –
 Field Data
- Evaluate Significance of A Severe Ground Exposure to Flight; i.e.
 Acceleration Factors (This Briefing)
- - ASTM B117
 - GM9540
 - G2





Overview

- A Large Flight and Ground Exposure Study Is Being Conducted To Evaluate Non-Chrome Paint Stackups In Comparison To Standard Chromated Systems
- The Emphasis is Corrosion as Produced By a Variety of Physical, Chemical, and Biological Stresses
 - Atmospheric/Coastal/Daytona Beach This Briefing
 - Flight on OML of military aircraft This Briefing
 - Fungal
 - Decon Solutions
 - Biocides
 - Urine
 - Wash Intervals
 - Wash Intervals + Biocides



Scope and Approach In Study

- >60 Paint Systems Under Evaluation on Ground At Battelle's Daytona Beach Exposure Site
 - Mix and match pretreatments, primers, and topcoats
 - Painted and Scribed Corrosion Sensors (Electrical/Quantitative)
 - Painted and Scribed 2024 T3 Aluminum Panels (Visual/Subjective)
 - Painted and Scribed Galvanic Couples (Steel/Al) (Visual/Subjective)
- 9 Paint Systems Being Flown on Painted and Scribed Corrosion Sensors On Upper Fuselage of 42 Aircraft (Initiated by OSD)
 - C130
 - C5
 - HC144



Examples of Ground and Flight

Objective



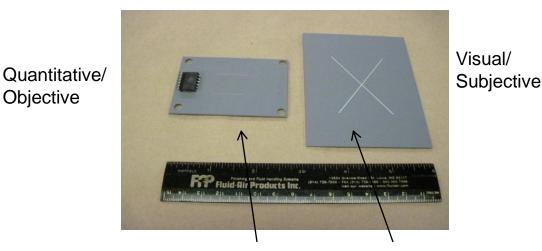
Daytona Test Rack @ 50 Meters (1 of 3 in use; ESI Steel = 165,000 microgm/cm2/yr.



Galvanic Samples; 1 Year; Ground



Panel with 6 sensors on C130; Upper Fuselage



Typical sensor and Panel

BUSINESS SENSITIVE



Examples of Appearance of Field Samples



Best NC/NC Coating 4+ Years; 2024 Al



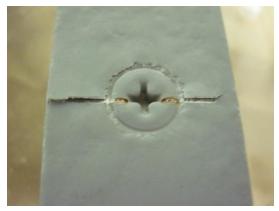
Typical Chromate Coating 4+ Years; 2024 Al



Poor NC Coating <1 Year; 2024 Al



Galvanic Test Samples < 1 Year; 2024 Al/Steel



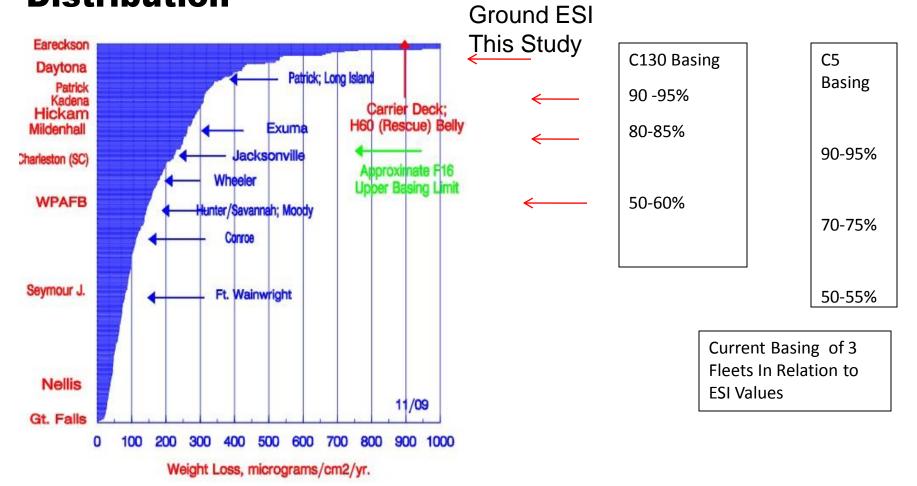
Galvanic with Poor NC Coating; < 1 Year



KC135 Wing Skin 2+ Years; C/C and NC Coatings



Current Fleet Basing vs Worldwide Environmental Severity Index (ESI) Distribution



Ground Severity = Accelerated Exposure vs. Most of US Military Land Basing



MIL-PRF-85285 Ty IV MIL-PRF-85285 Ty I

MIL-PRF-23377 CI C2 MIL-PRF-23377 CI N

MIL-PRF-85582 CI N MIL-DTL-81706

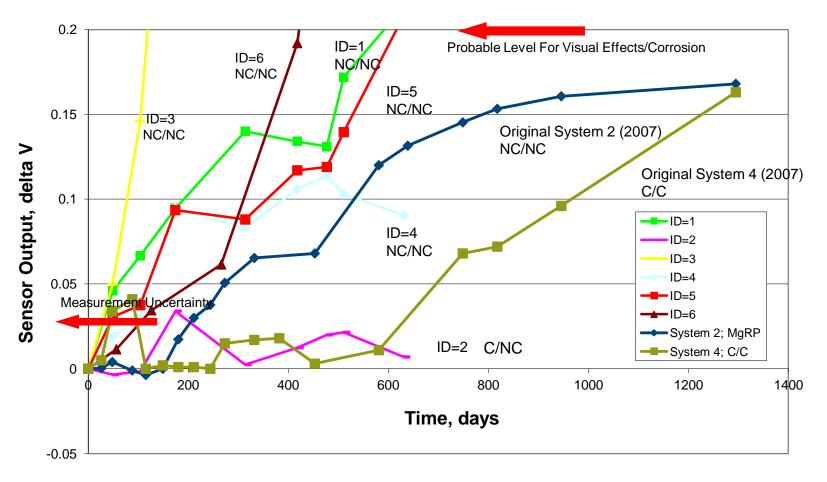
Paint Stackups For Flight and Ground Evaluations – 2007 and 2009 Start

2009 Long Term Start; Ground and C130, C5, HC144

- ID=1 Alumigold / 44GN098 (DEFT) / 99GY001 (DEFT)
- ID=2 Alumigold / AE2100 (XP406-110) ANAC / 99GY001(DEFT)
- ID=3 Alodine 5700 / 02GN084 (DEFT) / 99GY001 (DEFT)
- ID=4 Alodine 5700 / 16708TEP (Hentzen) / 99GY001 (DEFT)
- ID=5 Prekote / Sicopoxy 577-630 (ANAC) / 99GY001 (DEFT) C130 only
- ID=5 Prekote / 02Y040B (DEFT) / 99GY001 (DEFT)
 C5 only
- ID=6 DEFT 1015/3021 / 02GN093 (DEFT) / 99GY001 (DEFT) C130 ,HC144 only
- ID= 6a Prekote / 02Y040B (DEFT) / PPG Desothane HS Half of C5s only
- ID = 6b Prekote / 02Y040B (DEFT) / AE5000 (ANAC) Half of C5s only
- 2007 Long Term Start Ground and flight (H60/P3/AH64A)
- ID=7 Alodine 1200S / 02GN083 (DEFT) / 99GY001 (DEFT)
- ID=8 Prekote / AE2100 (first generation) / AE5000 (ANAC) Old System 2
- ID=9 Prekote / 02Y-040 (DEFT) / 03GY310 (DEFT)
 Old System 3
- ID=10 MIL-C-5541 / 02Y040 (DEFT) / 03GY310 (DEFT) Old System 4
- ID=11 Alodine 5200 / Sicopoxy / 03GY310 (DEFT)
- ID=12 Prekote / Americoat 3351 / 03GY310 (DEFT)

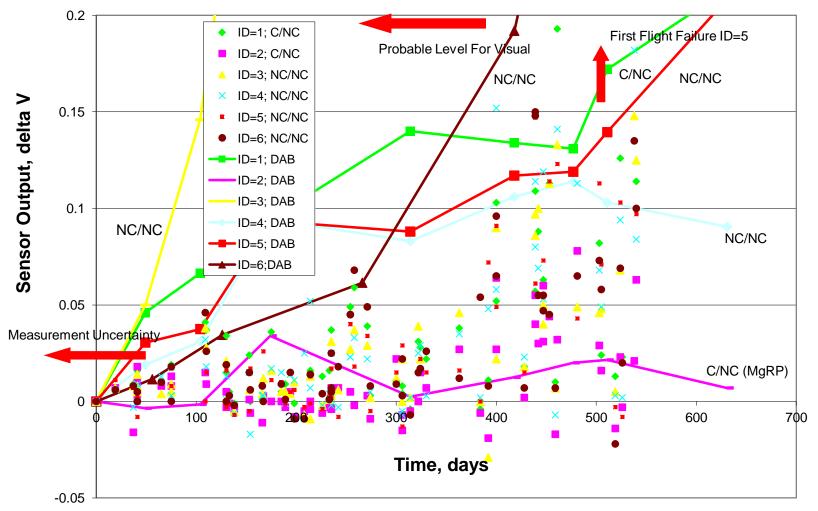


Summary of Sensor Results – Ground @ Daytona For Same Paint Systems As C130 Flight





Combined Data For Ground and On-Aircraft (Solid Lines = Ground; Points=Flight By T/N)



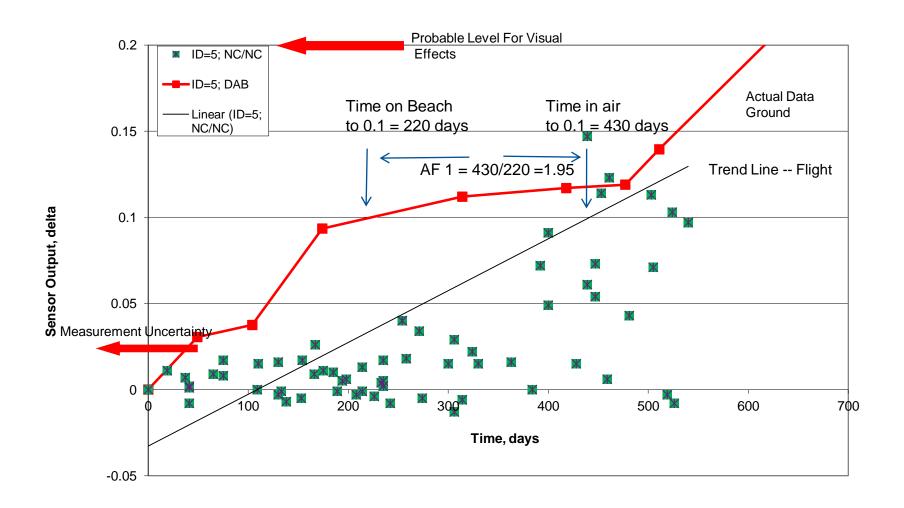


Significance of Data

- Paint Systems Are Degrading Faster In Severe Environment of Daytona Than On OML of Aircraft
- This Provides Some Degree of Realistic Acceleration = AF 1
- Painted aluminum on ground will take much longer to visual degradation than response of sensor = AF



Example of Acceleration Factor For Painted and Scribed Sensor In Flight vs. Daytona Beach Exposure -- One Paint System





Analysis of Ground vs. Flight Data

- Ground is Accelerated vs Flight Per Sensor Data
 - ESI on OML is less than at Daytona for worst case basing of probably all USAF fleets
 - Analysis does not apply to sea basing
- 2024 Al Panels (Visual) React Much Slower Than Sensors = Additional Acceleration For Sensors Data
 - Acceleration Factor Even Greater
- Same Order of Performance of Paint Stackups For Both – Sensor vs.Al (Gnd) and Sensor Gnd vs Flight
- Enough Data to Define Approximate Acceleration Factors



Acceleration Factors Gnd vs Flight and Gnd Sensor vs Gnd 2024 Al

 Sensors By Paint Stackup; Time (days) to Reach Same Electrical Change (Delta = 0.1 V)

)	Ground	Flight	Factor	Al (Visual)	Factor
- ID=1	180	400	2.2	670	3.7
- ID=2	>600	>700			
- ID=3	120	461	3.8	>720	>6
- ID=4	170	420	2.3	710	4.1
- ID=5	170	322	1.9	680	4
- ID=6	310	461	1.1	560	1.8
 MgRP (200) 	7) 670	> 1500		>1500	
Cr/Cr (2007) 980	> 1500		>1500	
		»	Av = 2.3		



Acceleration Factor (Cont'd)

- Painted and Scribed 2024 Al Panels Require 2-4
 Times Longer To Reach Visible Degradation
 Compared to Sensor Results
- Conservative Estimate = 4:1 Acceleration Factor For Severe Ground (Sensor) vs. Flight (2024 AI) For Worst Case Land Based Assets
- Significance of Sensor Results vs Painted 2024
 - Systems Deployed In 2007 Exposed to Equivalent of >10
 Years On Wing of C130 and C5
 - Large Matrix Started 2009 Now Exposed to Equivalent of ~8 Years On Wing of C130 and C5



Conclusions

- Procedures and Protocols Established For Flying Multiple Painted Corrosion Sensors on OML of Aircraft
- Approximate Acceleration Factors Now Established To Estimate The Meaning of Ground Based Exposures to Current ESI Levels of Fleet Basing
- Data Are Giving Optimistic Appraisals For Long Term Corrosion Protection For The Best Non-Chrome Systems For USAF Aircraft.
- Studies Are Continuing To Define Acceleration Factors More Precisely